## RFC Dallas, Inc. AIRCRAFT QUESTIONNAIRE

## Archer-II Version 1.0 (05/01/2025)

lot Name: Date:		Date:
Aircraft Registration#:	Model:	Serial#:
Answer the following questions by using the informacturent Weight and Balance supplement, placards affalM. After being reviewed by a Club Checkout Instructure. Safety Officer before solo flights may be conduct	fixed to the aircraft, the airc ctor, this questionnaire mus	craft Checklist, and the FARS &
1. How many fuel drains must be checked d	uring preflight?	
2. The Airplane Flight Manual states that the quarts.	e oil capacity of the O-3	360 series engine is:
(However, RFC recommends adding a quart of oil w	hen the level on the dipstic	k indicates less than 6 quarts.)
3. After starting the engine, the maximum t indicated is seconds.	ime that may elapse be	efore oil pressure must be
4. During the pre-takeoff engine run-up, the magneto is individually selected, the max difference between Left and Right mag is	imum allowable drop is	
5. To shorten takeoff distance, a flap setting	g of degrees r	nay be used.
For questions 6 thru 10 below, use the following crite subsequent questions pertaining to the aircraft's perj		narts in the AFM, to answer the
Conditions:		
Preflight Pressure Altitude	1600 ft.	
Temperature	30° C	
Aircraft Gross Weight	2400 lbs.	
Runway Surface	Paved/Level/Dry	
Wind	Calm	
6. Minimum aircraft runway <u>takeoff</u> distance i	sfeet.	
7. Using the recommended flap setting, a to required to clear a <u>50 foot obstacle</u> .	otal <u>takeoff</u> distance of	feet is
8. Assuming a 50 foot obstacle at the approuse a minimum <u>landing</u> distance of		, the aircraft will
9. The expected climb rate is fi	eet ner minute	

10. What is the power off stall speed, for the following conditions:				
, , , , , , , , , , , , , , , , , , ,	nd 0° angle of bank? nd 45° angle of bank?	MPH. MPH.		
11. The V-Speeds for	this aircraft in MPH a	re (complete the chart b	•	
	Vne	Vsi		7
	VIIE	VSI		
	Vno	Vso		
	Va	Maximum Glide (engine out)		
	Vfe	Enroute Climb		
	Vx	Final Approach (40 Deg Flaps)		
	Vy	Balked Landing (40 Deg Flaps)		
13. The maximum no	etrue airspeed when under the second when the second w	veight for this <u>actual</u>	<u>aircraft</u> is	lbs.
	Date on Wt&Bal			
	Basic Empty Wei (lbs)	ght		
	Arm (inches)			
	Moment (lb inch	es)		
15. At maximum Gro	ss Takeoff Weight, sp	ecify the following:		
	Forward CG Limit	(inches)		
	Rearward CG Lim	it (inches)		
16. With <u>maximum f</u>	uel on board how mud	ch additional weight	may be ca	arried aboard the
17. What is the maxing	mum weight for the b	aggage area?	lbs.	

## For questions 18 and 19 Sample questions, use the **following information**:

Basic Empty Weight (lbs)	1424.0
CG (inches)	86.14
Moment (lb-inches)	122,664.8
ARMS:	
Front Seats (inches)	85.5
Rear Seats (inches)	118.1
Fuel (inches)	95.0
Engine Oil (inches)	31.7
LOADING CRITERIA:	
Pilot (lbs)	185
Co-Pilot (lbs)	190
Rear Passengers (lbs)	105
Fuel (gals)	48

	Tuci (gais)	40	
18. Results ar	e:		
a) The	Gross Takeoff Weight (lbs) is	S:	
•	Takeoff C.G. (inches aft of da		-
•	e aircraft loaded within allow		( Yes / No )
· ·	ne aircraft loaded within allo	<del>-</del>	
19. Given the	loading scenario from the pr	revious question, addi	ng 50 lbs. of weight
	ompartment (Arm = 142.8) will	-	5
2.08.00			
a) Unda	ated Gross Takeoff Weight (lbs):		
, ,	3 , ,		
b) Upaa	ated Takeoff C.G. is (inches aft of c	aatum):	
This added	d weight will cause <i>(circle all ti</i>	hat apply):	
a\ Tb.a	a aimound the manufacture of Continuing	:+ +	
· ·	e aircraft's rearward C.G. limi		
•	e aircraft's forward C.G. limit		
	e aircraft's maximum gross w	<del>-</del>	
f) The	aircraft to be within weight	and C.G. limits.	
20. <u>Usable</u> fu	el capacity is: TOTAL	gals; To the TABS	gals.
21. The rated	BHP of the engine installed i	in this aircraft at maxi	mum allowable RPM
	_BHP at RPM.		

22. Proper tire inflation pressures (psi) are: Mains: \_\_\_\_\_, Nose Wheel: \_\_\_\_\_.

23.	Can the ELT be activa	ted manually? ( Yes / No )
24.	This aircraft is equipped	d with a heated pitot tube? ( Yes / No )
25.	* *	ing indicator is installed in this aircraft <i>(circle all that apply)</i> ? uzzer / Light / Siren )
26.	How do you test the s	call warning system on the ground?
27.	The battery is ( 6 / 12	/ 24 ) volts.
28.	Which of the following ( Pitch / Ro	axis can the pilot trim <i>(circle all that apply)</i> ? oll / Yaw )
29.	What initial actions sho flight?	uld immediately be taken upon losing engine power during <u>cruise</u>
	a)	
	b)	
	c)	
	d)	
	e)	
	f)	
	g)	
30.	What are the prescribe developed spin?	d aircraft control inputs to initiate a recovery from a fully
	Power	
	Ailerons	
	Rudder	
	Elevator	

31. This aircraft is approved for flight into known icing conditions? (Yes / No)
32. The GPS in this aircraft is certified for IFR operations? (Yes / No)
33. The maximum CHT temperature is F. Target CHT temperatures should be at or below F.
34. According to FAR 91.7, the is responsible for determining that the aircraft is a condition for safe and legal flight.  +++++++++++++++++++++++++++++++++++
35. How do you enable/disable the electric trim
36. If the turn coordinator is inoperative, thesystem is also inoperative
37. How do you check the backup battery on the Garmin G5?
38. The HSI is powered by a vacuum pump? ( Yes / No )
39. The aircraft is equipped with the S-Tec Autopilot System Model ( $10 \ / \ 20 \ / \ 30$ )?
40. The autopilot can fly ILS and LPV approaches with vertical guidance? (Yes / No)
41. The autopilot can auto-intercept the inbound LOC, GPS or VOR course? ( Yes / No )
42. What in-flight steps are necessary prior to engaging ALT HOLD mode?  (a), (b),  (c), (d),
43. What does the RDY light indicate on the turn coordinator?
44. When should the pilot select HI-TRK mode during an LOC, GPS or VOR approach?  (a), and (b)
45. How will the pilot be notified when the aircraft is "out of trim" (autopilot ON)?  (a), and (b)
Pilots (Printed Name & Signature):
Reviewed by (Printed Name / Signature): Date: